

Generic Mission Architectures

benefiting from these instruments

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Overview

- Definition & examples

Mission Architecture?

"mission architecture" - Google Search - Windows Internet Explorer provided by European Space Agency

http://www.google.co.uk/search?hl=en&q=%22mission+architecture%22&bav=on,or,r_gc,r_pw,r_qf,cf.osb&biw=1536&bih=650&wrapid=tlif133977719016210&um=1&ie=UTF-8&tbm=isch&source=og&sa=N

Google "mission architecture" Sign in

Search About 17,700 results (0.49 seconds) SafeSearch moderate

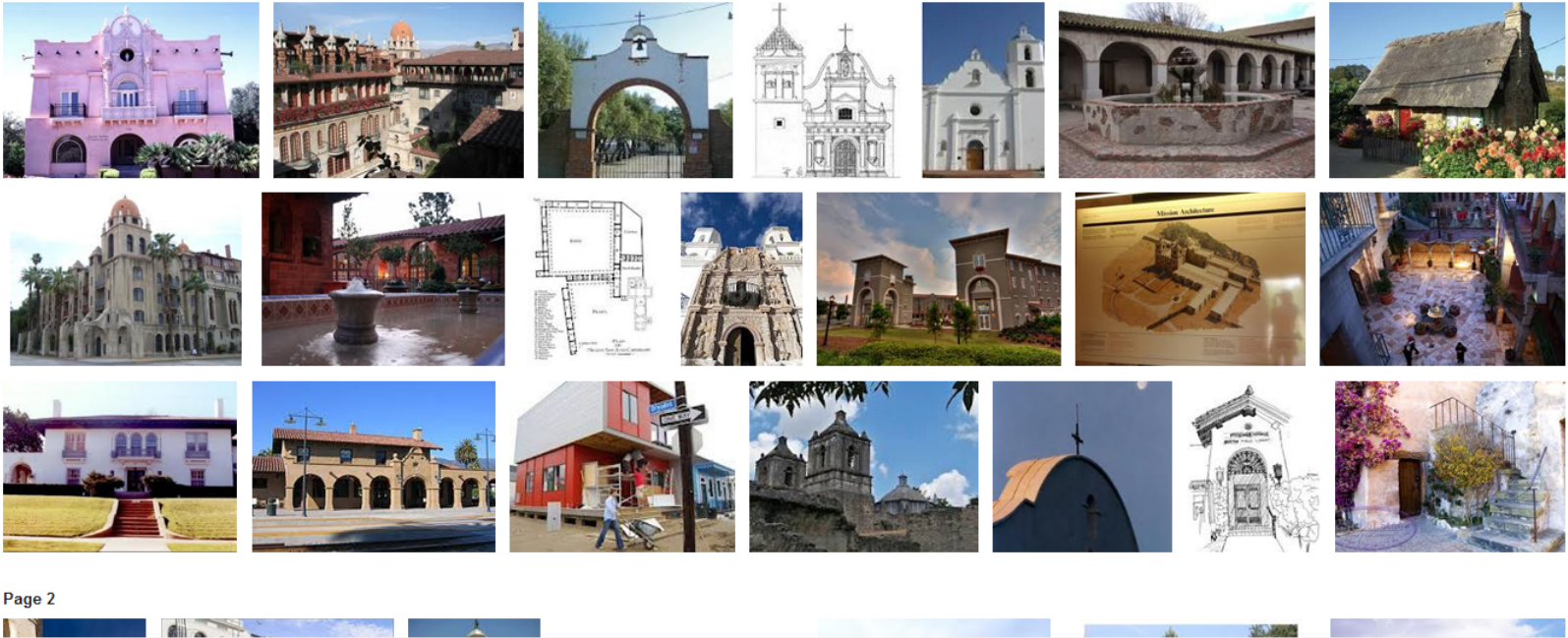
Web
Images
Maps
Videos
News
Shopping
More

Any time
Past 24 hours
Past week
Custom range...

All results
By subject

Any size
Large
Medium
Icon
Larger than...
Exactly...

Page 2



The grid displays a variety of mission architecture. It includes a pink building with a central tower, a large stone building with multiple windows, a blue arched gateway, a white church facade with a cross, a white church with a bell tower, a stone courtyard with arches, a thatched-roof building, a large stone building with a central tower, a stone courtyard with a fountain, a stone building with a central tower, a stone building with a central tower, a stone building with a central tower, a stone building with a central tower, a stone building with a central tower, a stone building with a central tower, a stone building with a central tower, a stone building with a central tower, a stone building with a central tower, a stone building with a central tower.

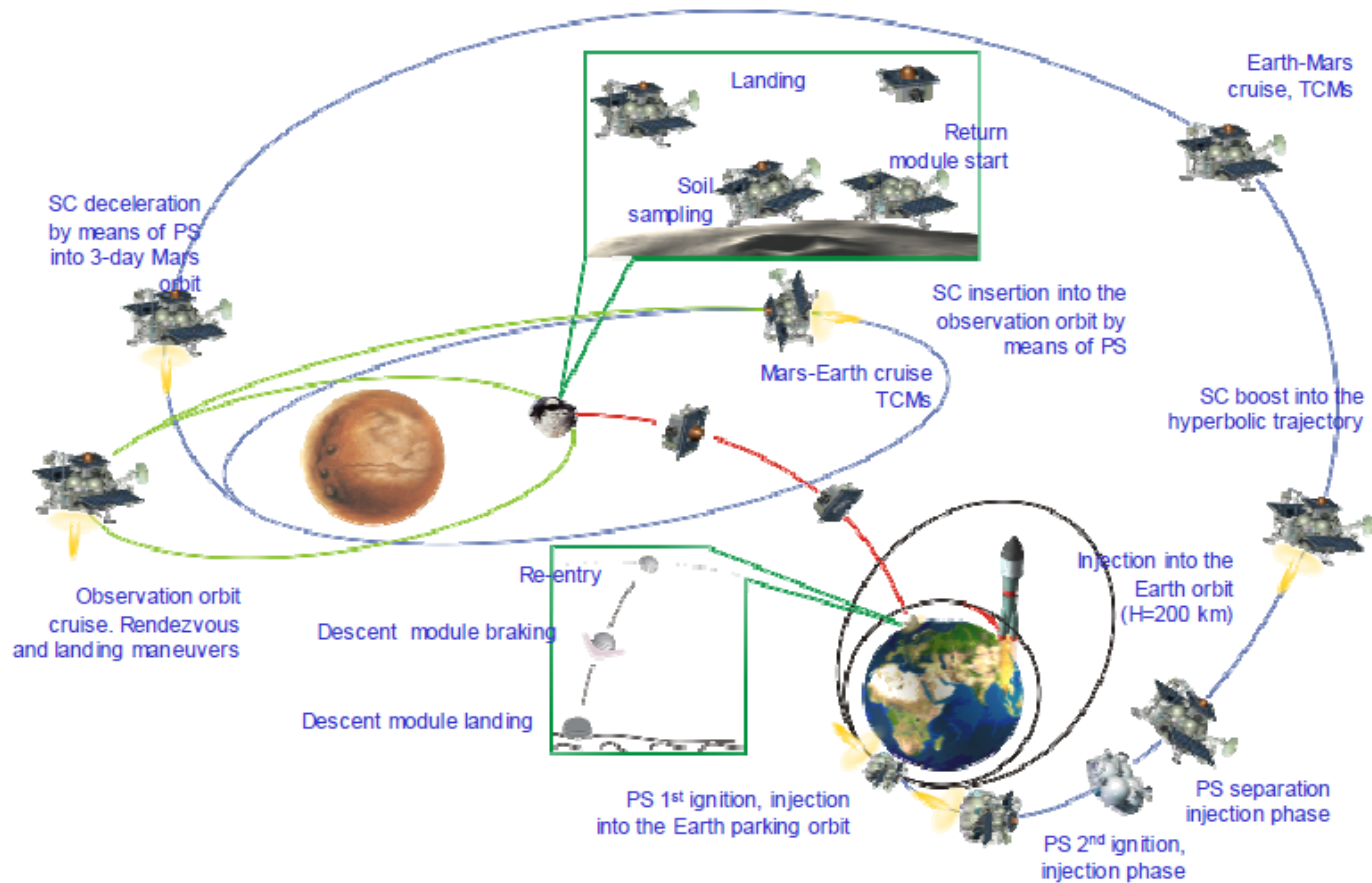
What is a (Probe) Mission Architecture? (1)

- A top-level* description of the spacecraft elements and their relation to each other, the mission target and Earth during mission operations
 - *May be part of a higher-level programme architecture
 - Devised as a means to address a set of mission objectives...
 - ...or alternatively to show how a set of heritage elements may be used.
 - Can be described as part of the mission profile

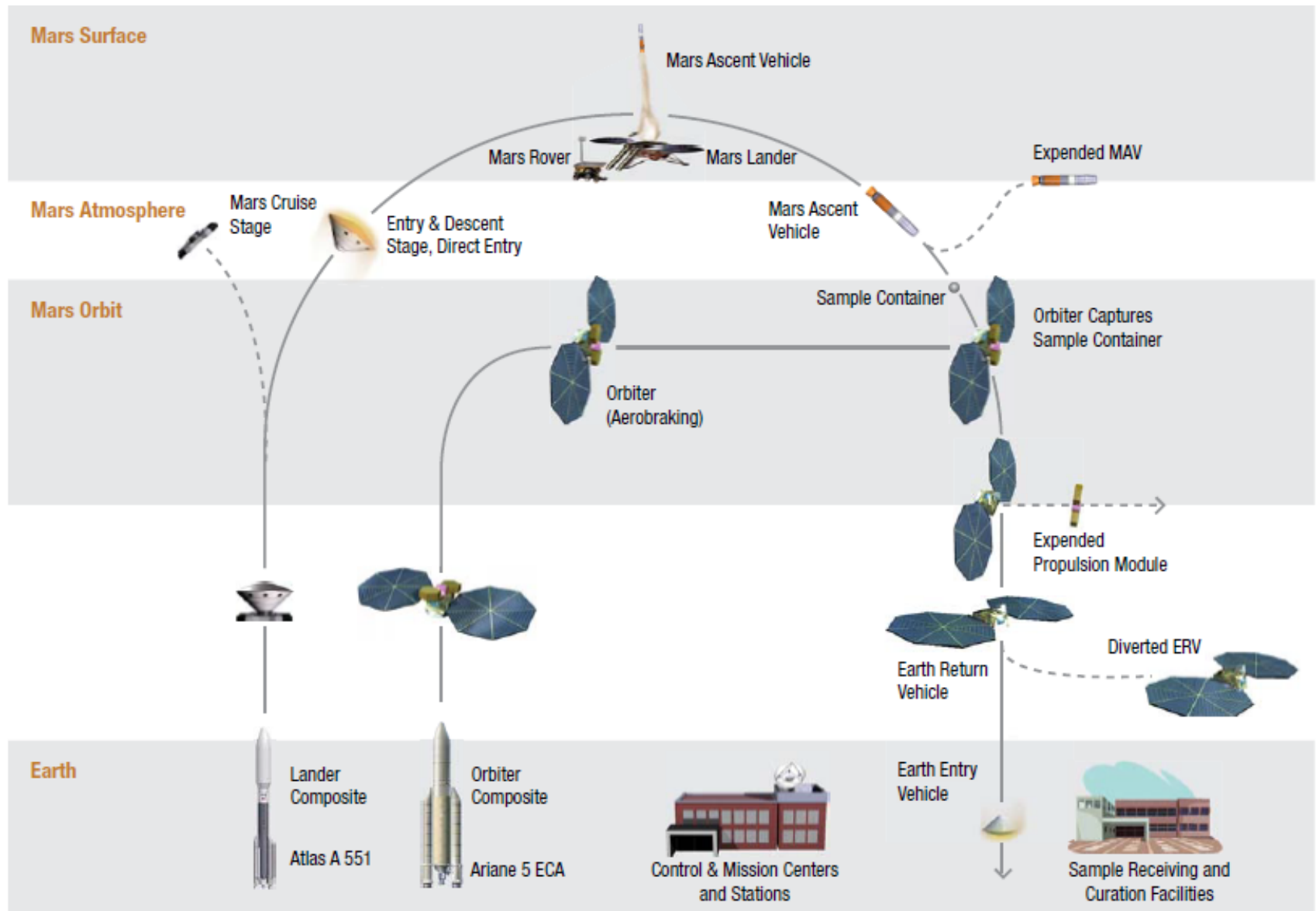
What is a (Probe) Mission Architecture? (2)

- The mission architecture indicates:
 - the main events of a mission timeline
 - E.g. launch, orbit insertion, separation, landing,...
 - configuration of the spacecraft
 - E.g. launch configuration, orbiter or carrier, number of probes, ...
 - configuration of key systems
 - E.g. power source, EDL architecture
 - often also the communications scenario
 - Data relay vs. Direct To Earth (DTE), ground segment
 - May or may not show the orbital geometry...
- (There's probably no such thing as a generic mission architecture!)

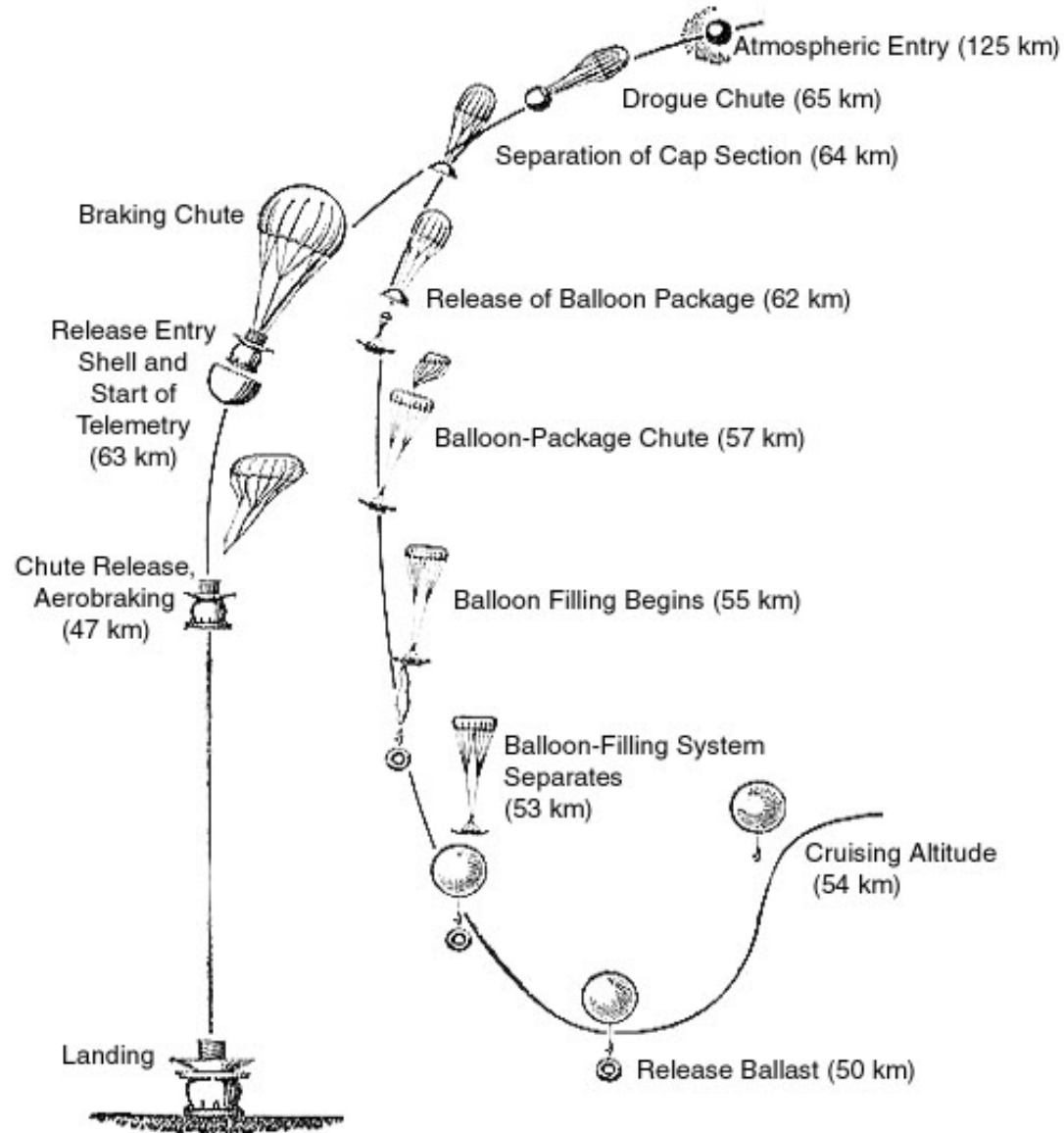
PHOBOS SAMPLE RETURN MISSION PROFILE



Sample Return Architecture

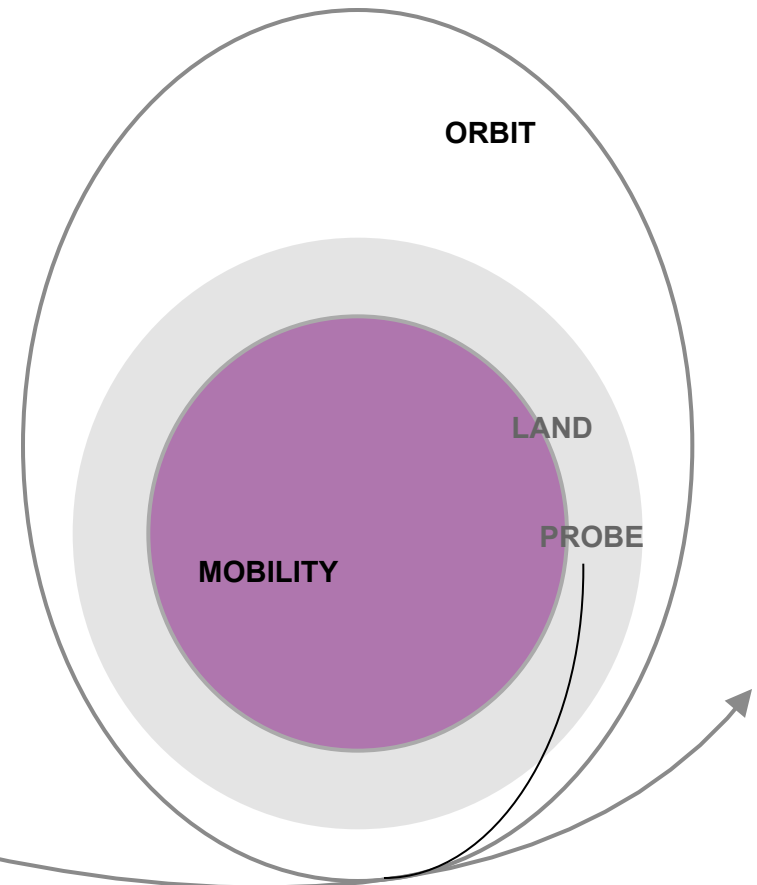


EDL & EDI architecture



Mission Architecture vs. Complexity

- Ground-based
 - telescopes, radio science, laboratory, analogues, meteorite & sample analyses
- Flyby
- Destructive impact / [entry probes](#)
- Orbit
- [Atmospheric entry probe](#)
- [Lander](#)
- [Mobility \(rover, balloon, etc.\)](#)
- [Sample return](#)
- (Human exploration)



FLY-BY

SAMPLE RETURN

Some Benefits of *In Situ* Investigation

- Measurements that are impossible remotely
 - Many environmental parameters, physical properties
 - Detailed composition (trace el'ts, isotopic,...), petrology
 - Measurements hindered by the atmosphere
- Validation of measurements or inferences from remote investigations, modelling, lab simulation, etc.
- Scales much finer than achievable remotely
- Interaction with the environment (active techniques)
- Sampling of material for *in situ* analysis
- Access to sub-surface (sampling, thermal meas., etc.)
- Mechanical coupling for seismology & rotation meas.
- Sampling for return to terrestrial labs

Single or Multiple probes?

- Network measurements
 - Meteorology
 - Seismology
 - Geodesy
- Can be used for redundancy (though rarely since mid-80s)
- Multiple probes on a single craft – likely to mean ballistic entry and large landing ellipse, which luckily is OK for network experiments
- Multiple probes demanding on data relay architecture
 - covering 4 probes at the same time as orbit insertion practically impossible!

Probe Delivery

- Atmospheric probes to: Venus, (Earth), Mars, Jupiter, Saturn, Titan, Uranus, Neptune
- Generally from a carrier spacecraft of some sort
 - Earth's Moon (and Earth) are the only possible exceptions
- Delivery from hyperbolic, highly elliptical or near-circular orbit
 - Together with the target's escape speed and rotation speed, the geometry of this choice determines the entry speed relative to the atmosphere
 - And together with the Flight Path Angle (FPA) and other parameters helps determine what ballistic coefficient, aeroshell design, Thermal Protection System (TPS) to select
- The probe may have the correct velocity already after separation
 - If not, probe needs its own deorbit propulsion.
- Probe may be on course for entry
 - Avoidance manoeuvre if carrier needs to survive for orbital mission and / or data relay
- Importance of atmospheric models (profiles of density, wind, variability,...)
- And of surface datasets (elevation, slopes, thermal inertia,...)
- Landing ellipse – combined effect of the various dispersions

Spin

- Spin may be needed for stability and thermal reasons during coasting phase
- May be imparted by parent spacecraft or separation mechanism
- May be too fast
 - yo-yo despin
- May be too slow
 - Spin-up thrusters
 - Spin vanes
 - Oscillation dampers
- If guided entry then needs to be zero
- Useful for stability (and heat distribution) during ballistic entry

Entry

- Pre-entry environment
 - Fields, radiation
- Entry state
 - Navigation problem
- Probe dynamics – IMU
- Engineering instrumentation
 - Studies response of system to the environment but also needs to know what the environment is
 - Gas properties, heat fluxes, TPS response
- Science instrumentation
 - Aims to derive properties of the target environment independent of probe
 - Probes in the flow, upper atmospheric composition

Descent

- Mars descent very short (~2min) and mission-critical so difficult to add experiment objectives
- Venus, Earth, Titan, Giant planets – medium to long descent
- For giant planets and (most of) Huygens, the descent *is* the mission
- Many more measurement opportunities than entry:
 - Probe dynamics (more complex than entry, however)
 - Radio science
 - Local
 - Integrated (up or down)
 - Surface – imaging, radar
 - Analytical measurements
 - Sampling and analysis
- Measurements have to be compatible with the descent dynamics and environment
- Some may require sophisticated IF to the environment – needs to be compatible with the platform

Phased Descent Measurements

- Hostility of Venus lower atmosphere may mean some in situ measurements are only possible higher up ($>30\text{km}$) and one accepts loss of part of the payload
- i.e. tailor payloads to different altitude ranges or surface

DTE vs. Relay

- DTE used nowadays usually for EDL monitoring only
- DTE may however be attractive for some missions where the objectives can be achieved with the data rate / volume available and establishing a data relay is too costly for the envelope
- Mission architecture has to achieve geometry for relay
- Deep probes – watch out for absorption

Power Architecture

- Primary battery
 - Suited to short / limited duration missions
- RTG + secondary battery
 - Suited to cold targets, e.g. Mars and Titan
- Solar array + secondary battery
 - Suited to longer missions in inner solar system: Mars surface, maybe Venus balloons?

Targeting

- The need to target a particular location (small area for landing ellipse) may drive the mission architecture
 - Guided entry instead of ballistic
 - Rover to reach the target from the landing site

Duration

- How long does it take to carry out the desired objectives of the mission in the atmosphere / on the surface?
- Major effect not just on any consumables but also power and thermal architecture, e.g. 'keep alive' heaters or cold electronics with a limited number of cycles?

Preprogrammed vs. commanded

- All probes preprogrammed to surface – the long descent destinations are too far away to make commanded operation attractive
- Long-lived balloons would be a possible exception
- Interaction almost always desirable once probe is on the ground

Programmatics & Mission Architecture

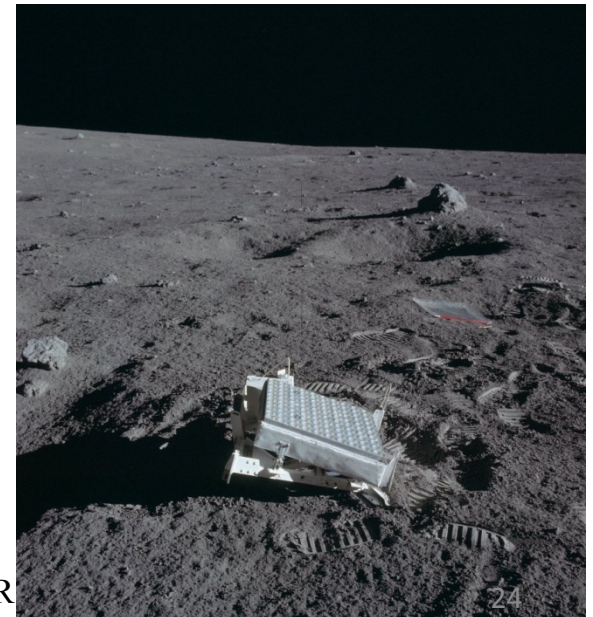
- What is to be classed as science payload?
 - Usually selected by AO
- What is provided by the ‘system’?
 - E.g. system data, deployments, sampling,...
 - Can’t count on science return but there often is one, even if not guaranteed
 - (Check auxiliary information, data rights, archiving)
- What needs to be in there at the top as a programmatic objective?
 - E.g. tech demo, data relay for other missions, compatibility with other sample return elements...

Measurement Categories

- Geodesy / tracking
- 'Space physics'
- Atmospheric physics profiles
- Aerosol physics
- Fluid physics
- Sub-surface sounding
- Surface physical properties
- Imaging
- Photometry & spectrometry
- Microscopy
- Chemical composition
- Mineralogical & elemental composition
- Sample analysis
- Sample return

Geodesy / tracking

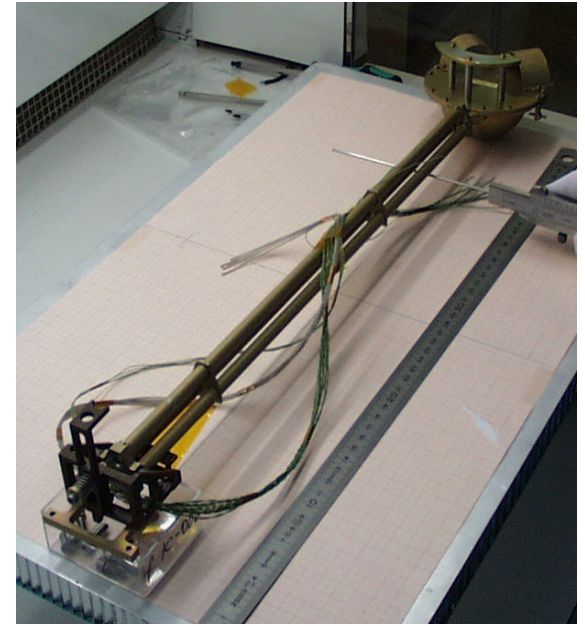
- Radio science (Doppler ranging, VLBI)
- Laser geodesy (retroreflectors)
- Radar/laser ranging/altimetry
- Gravimetry (link to seismometry)



Apollo LRRR

‘Space physics’

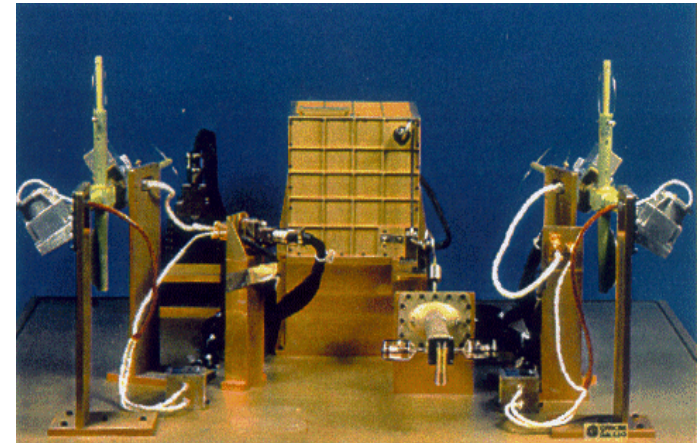
- Radiation
- Magnetometry
- Potentially also electric field, plasma sensors



ROMAP

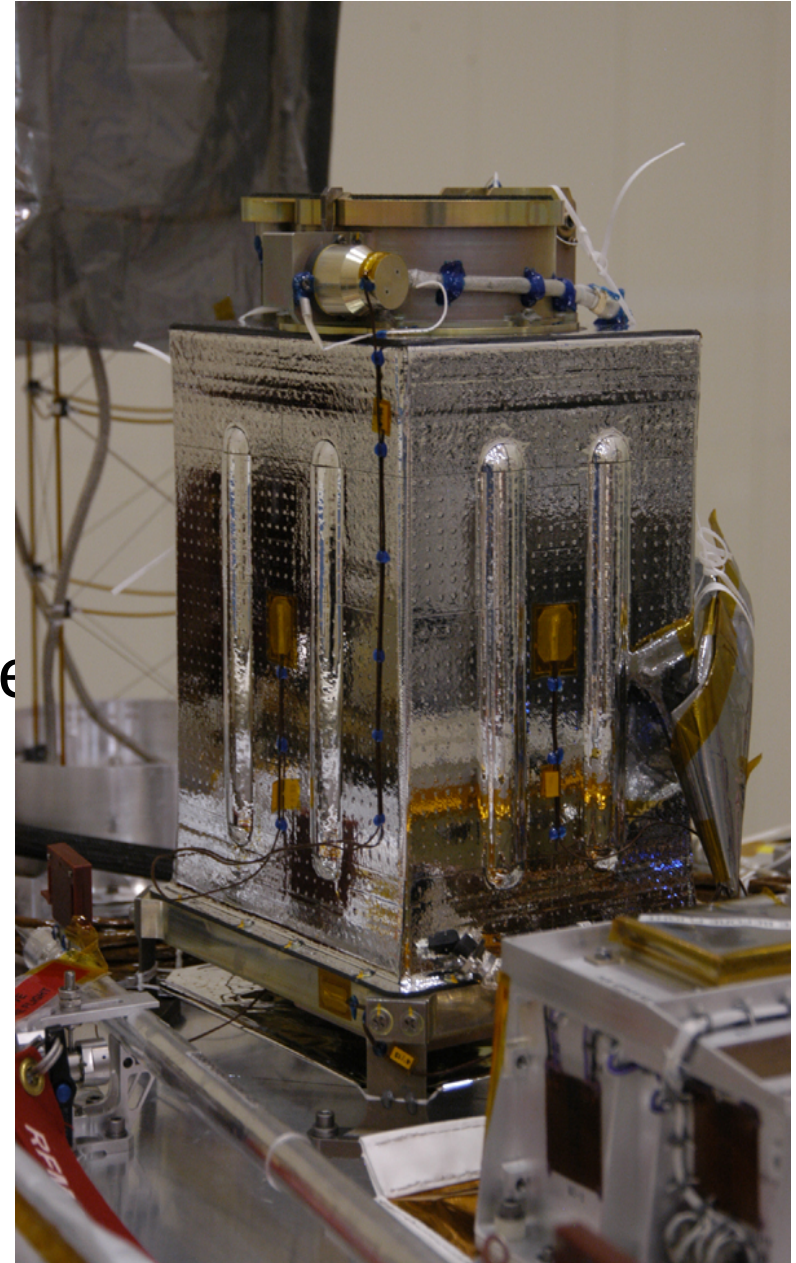
Atmospheric physics profiles

- Density (entry accelerometry)
- Temperature
- Pressure
- Humidity
- Wind / Turbulence
- Electrical field / conductivity / waves



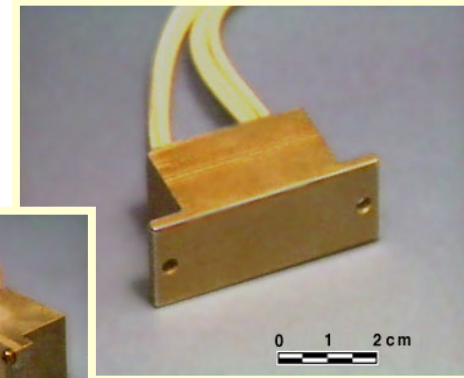
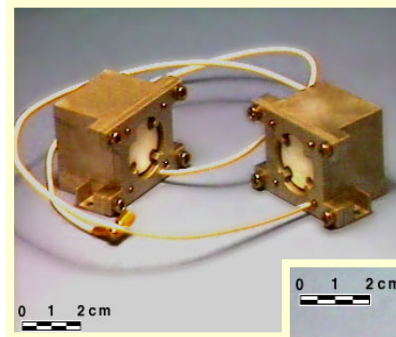
Aerosol physics

- Dust detector
- Nephelometry
- Cloud particle size spectrometer
- LIDAR (for clouds)

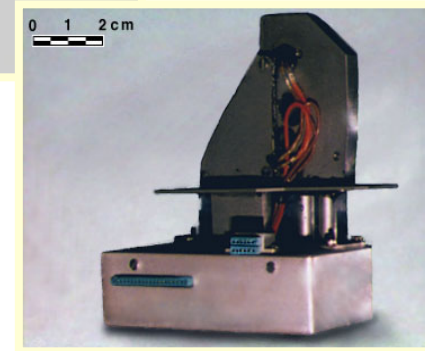


Fluid physics

- Refractive index
- Density
- Motion (waves)
- Speed of sound
- Microphone
- Electrical conductivity / permittivity

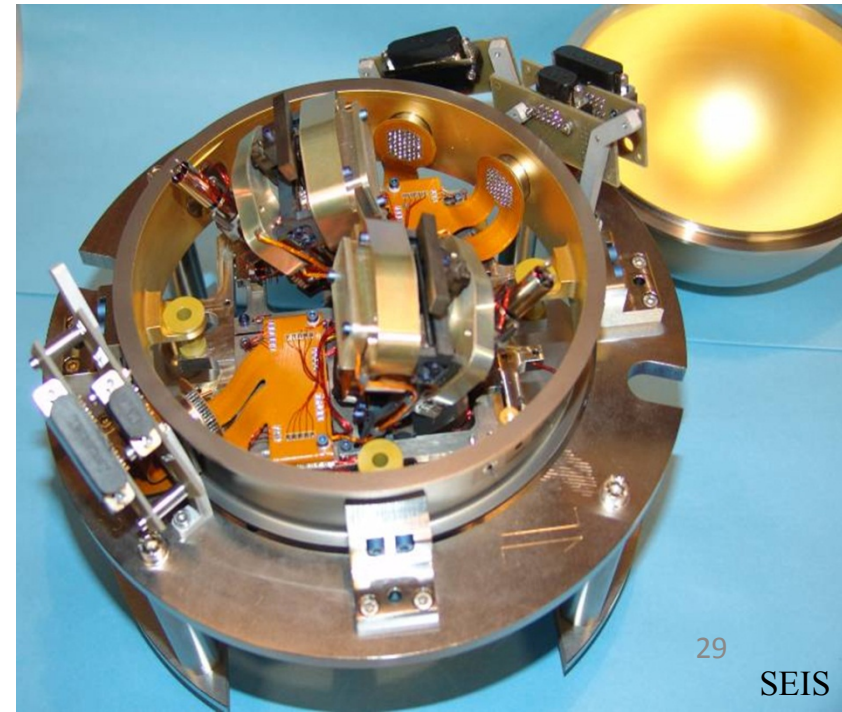


SSP



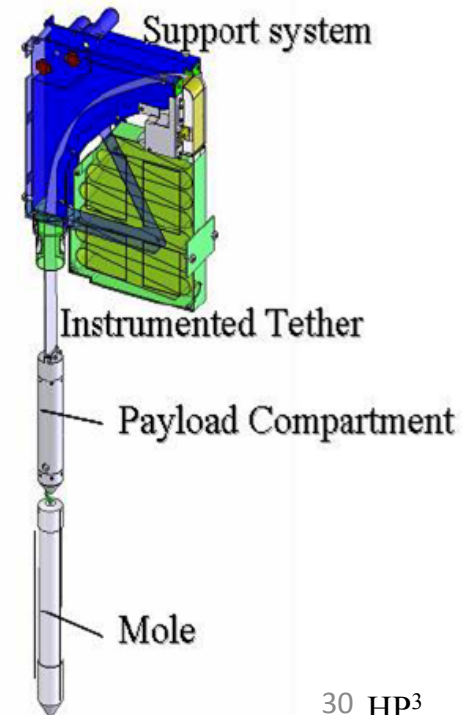
Sub-surface sounding

- GPR
- Radio Reflection Tomography
- Radio Transmission Tomography
- Sonar
- Seismometry (active & passive)



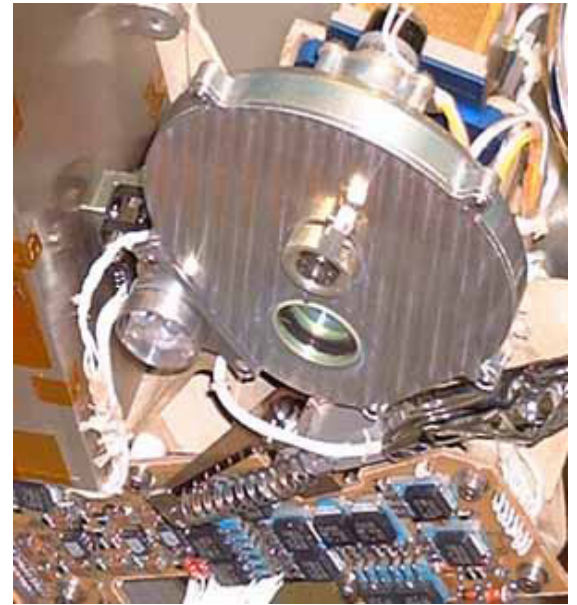
Surface physical properties

- Temperature / profile / conductivity / diffusivity / heat flow (IR radiometry, T sensors)
- Magnetic permeability / susceptibility
- Electrical Permittivity / conductivity
- Abrasion
- Adherence
- Landing dynamics / attitude
- Geotechnical Properties
 - Penetration resistance
 - Drilling resistance
 - Trafficability
 - Bulk density



Imaging

- Cameras
 - Descent
 - Panoramic (wide-angle)
 - Narrow-angle
 - Close-up
- Filters, polarisation, etc.
- Imaging of passive elements e.g. windsocks, sundial, colorimetric patches, magnets, etc.
- Imaging spectroscopy (Vis, near-IR to thermal IR)



SCS

Photometry & spectrometry

- Atmospheric profiling
 - Upward-looking
 - Downward-looking
- Atmospheric transmission to surface
- Temporal variation
- Spectral resolution (e.g. UV)
- Surface mineralogy
 - Point spectrometer
 - Raman

UVIS



Microscopy

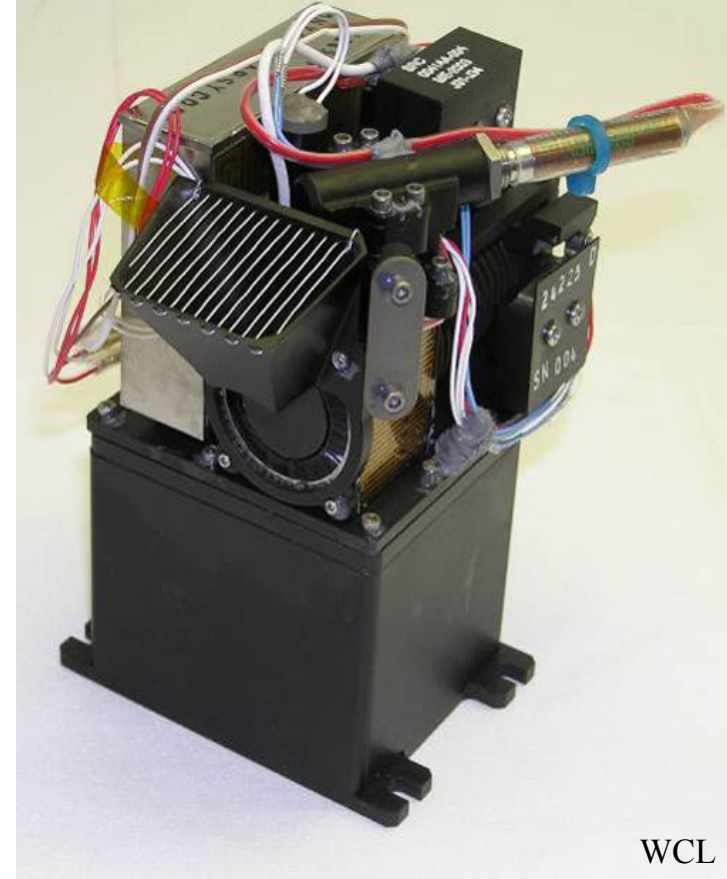
- (Imaging) IR Microscope
- Scanning Electron Microscope
- Atomic Force Microscope



ÇIVA-M

Chemical composition

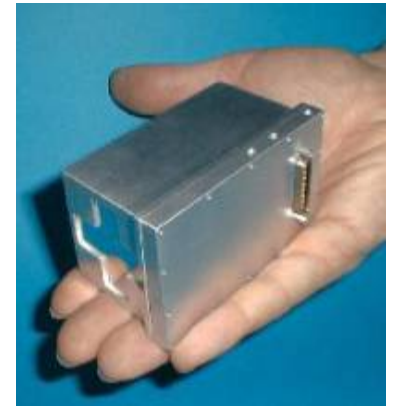
- Chemical gas analyser
- Geochemical indicator
- Oxidation sensor
- Helium abundance detector
- Evolved water experiment (e.g. thermogravimetry)
- Wet chemistry lab
- Tuneable diode laser spectrometer
- Neutral Mass Spectrometer



WCL

Mineralogical & elemental composition

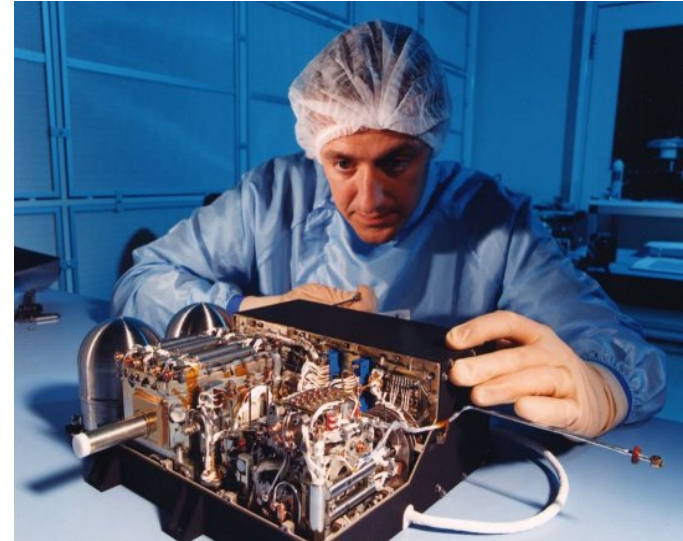
- X-ray diffractometer
- X-ray Fluorescence Spectrometer
- Alpha-(Proton-)X-Ray-Spectrometer
- Laser Ionisation Breakdown Spectrometer
- Mößbauer spectrometer
- Neutron spectrometer / pulsed neutron source
- Gamma ray spectrometer



MIMOS II

Sample analysis

- GCMS
- Aerosol collection / pyrolysis
- Evolved gas analyser
- ...



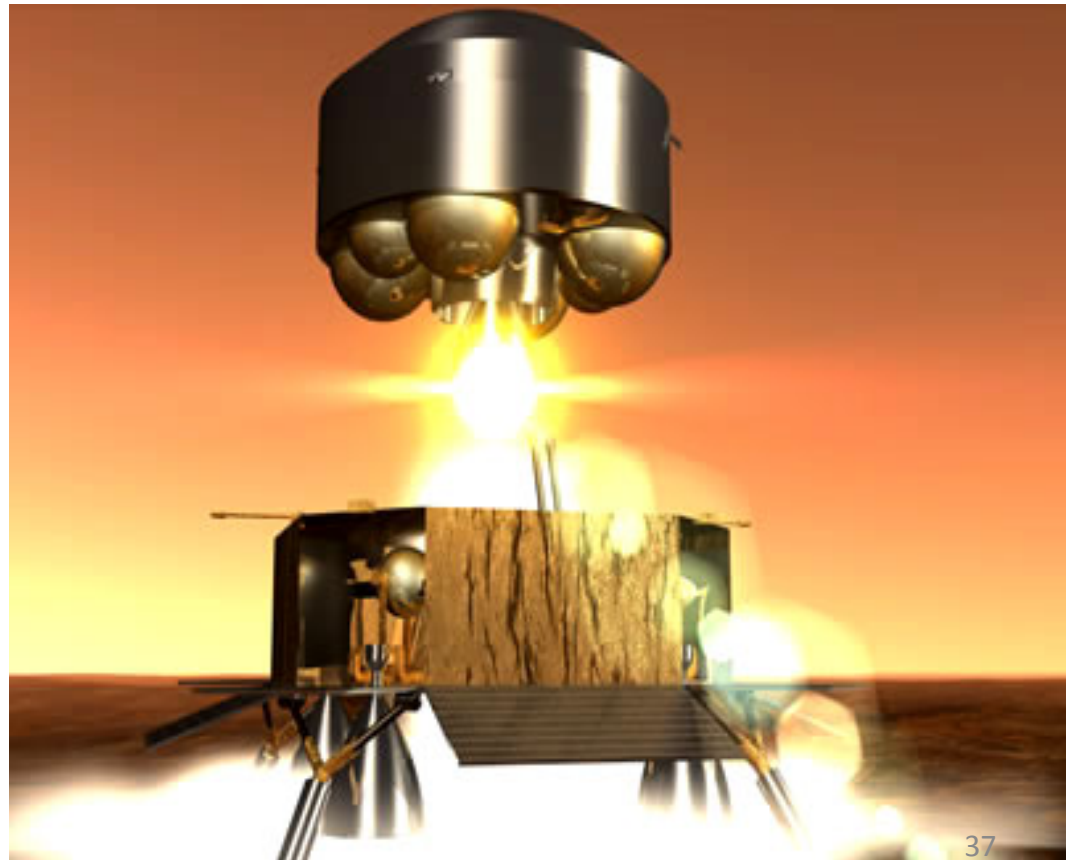
Rosetta - Ptolemy




Beagle 2 – Gas Analysis Package

Sample return

- Atmosphere
- Aerosols
- Dust
- Surface regolith
- Rock samples



Trends

- Network science (geophysics, meteorology)
- Astrobiology
- Mobility (across & below surface)
- Bulk properties  tracers
- Wider international participation / co-operation
- Lower frequency of missions
- Higher data rates / volumes
- More narrowly focused objectives
- Importance of landing site selection
- Sampling and sample return
- More exotic / extreme target environments
- Exoplanets / comparative planetology
- Money-limited
- Exploration context
- Establishment of programmes
- More precise measurements of the same targets / Focus on data quality
- Narrow objectives for smaller missions only

Payload Mass Fractions

